



The Stability of the Steady Motion of a Gravitating Gyrostat and Spheroid

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Abstract:

The paper investigates the stability of steady motions of a gravitating gyrostat interacting with a spheroidal rigid body under mutual Newtonian attraction. The gyrostat is modeled as a rigid body containing an internal rotor with constant spin, while the spheroid represents an axially symmetric mass distribution generating a central gravitational field with quadrupole effects. Using the equations of motion derived from the Hamiltonian formulation of rigid body dynamics, we characterize relative equilibria corresponding to steady translational-rotational motions, including uniform rotations and circular orbital configurations. The stability analysis is carried out by applying the energy momentum method and Lyapunov's direct method, supplemented by linearization around equilibrium solutions. Conditions for nonlinear and linear stability are obtained in terms of physical parameters such as the gyrostatic moment, mass distribution, oblateness of the spheroid, and orbital angular velocity. Special attention is given to the influence of the internal rotor on the stabilization or destabilization of steady motions. It is shown that gyrostatic effects can significantly enlarge the domain of stability and, in certain regimes, compensate for destabilizing gravitational gradients produced by the spheroidal field. Several limiting cases are discussed, including the reduction to a classical rigid body without internal rotors and the motion in a purely central gravitational field. The results provide a unified framework for understanding the coupled translational and rotational dynamics of gyrostatic systems in non-spherical gravitational environments. These findings are relevant to applications in astrodynamics and spacecraft attitude dynamics, particularly for satellites equipped with control moment gyros or reaction wheels operating near oblate celestial bodies.

Keywords: Gravitating gyrostat; spheroid; steady motion; relative equilibrium; stability analysis; rigid body dynamics; energy momentum method.

Introduction:

The study of rigid body motion under the influence of gravitational forces has occupied a central position in classical mechanics since the time of Newton, Euler, and Lagrange. From the earliest investigations of planetary motion to modern applications in spacecraft dynamics, the interaction between rotational dynamics and

gravitational fields has consistently revealed rich mathematical structures and complex physical phenomena. Among the most intriguing problems in this domain is the coupled translational and rotational motion of extended bodies subject to gravity, particularly when the bodies involved deviate from perfect spherical symmetry or contain internal mechanisms capable of storing angular momentum.

One such system is the gyrostat, a rigid body equipped with internal rotors or flywheels whose spins remain constant relative to the body frame. Gyrostats serve as fundamental models for a wide range of engineering and physical systems, including satellites with reaction wheels, control moment gyros, and natural bodies with internal angular momentum redistribution. The presence of internal rotors introduces additional conserved quantities and modifies the effective inertia of the body, thereby altering both the qualitative and quantitative behavior of its motion in a gravitational field.

Another class of systems of great relevance is that of spheroidal gravitating bodies. Many celestial bodies, such as planets and stars, are better approximated as spheroids rather than perfect spheres due to rotation-induced flattening. This deviation from spherical symmetry leads to gravitational fields that differ significantly from the classical central Newtonian potential, introducing higher-order multipole terms that influence the motion and stability of nearby bodies. The oblateness or prolateness of a spheroid can produce gravitational gradients that strongly couple with the rotational dynamics of an orbiting rigid body. The interaction between a gyrostat and a spheroidal gravitating body represents a natural and physically meaningful extension of classical rigid body problems. Such systems arise in astrodynamics, where artificial satellites with internal attitude control devices operate in the gravitational field of oblate planets, as well as in theoretical mechanics, where they provide a testing ground for advanced stability theories and Hamiltonian methods.

Historical Development of Gyrostat Dynamics

The concept of the gyrostat was introduced in the late nineteenth century as an extension of rigid body dynamics to account for internal angular momentum. Early contributions by Kelvin and Tait laid the groundwork for understanding gyroscopic effects in mechanical systems, while subsequent developments by Zhukovskii, Chaplygin, and Appell formalized the mathematical treatment of rigid bodies with internal rotors. In classical rigid body theory, the rotational motion is governed by Euler's equations, which depend solely on the body's inertia tensor and external torques. The introduction of a gyrostatic moment modifies these equations by adding constant vectors representing the internal angular momentum of the rotors. This modification leads to new integrals of motion and, in some cases, stabilizing effects that do not exist in ordinary rigid bodies. Throughout the twentieth century, gyrostats became increasingly important in engineering applications. The development of spacecraft attitude control systems brought renewed attention to the theoretical foundations of gyrostat dynamics. Reaction wheels and control moment gyros are, in essence, internal rotors that exchange angular momentum with the spacecraft body, enabling precise control of orientation without expelling mass. From a mathematical standpoint, gyrostats provided fertile ground for the application of modern analytical tools, including Hamiltonian mechanics, Poisson geometry, and perturbation theory. Researchers discovered that gyrostatic systems often possess rich symmetry structures, allowing the use of reduction techniques and energy-based stability criteria.

Gravitational Fields of Spheroidal Bodies

While early studies of rigid body motion in gravity often assumed a point-mass or spherically symmetric central body, it soon became evident that such simplifications

are insufficient for accurately describing real celestial environments. Rotating astronomical bodies tend to assume spheroidal shapes due to centrifugal effects, leading to mass distributions that are axially symmetric but not isotropic. The gravitational potential of a spheroid differs from the Newtonian potential of a point mass by the presence of higher-order terms, commonly expressed through zonal harmonics. The dominant correction is usually associated with the quadrupole moment, often denoted by the coefficient J_2 , which accounts for the body's oblateness. This term introduces a dependence of the gravitational force on the orientation and position of the orbiting body relative to the spheroid's symmetry axis. For rigid bodies and gyrostats in orbit, these gravitational gradients generate torques that couple the translational motion of the center of mass with the rotational motion of the body. Such coupling can lead to complex dynamical behaviors, including precession, nutation, and resonance phenomena. Importantly, the stability of steady motions is profoundly affected by the non-spherical nature of the gravitational field.

Motivation for Studying Steady Motions

In the analysis of mechanical systems, steady motions also referred to as relative equilibria play a fundamental role. These motions correspond to solutions of the equations of motion in which the system evolves in a simple, time-invariant manner when observed in an appropriate moving frame. Examples include uniform rotations, circular orbits with constant attitude, and synchronized translational–rotational motions.

Steady motions serve as reference states around which more general motions can be understood. Their stability properties determine whether small perturbations grow, decay, or remain bounded over time. In practical terms, stable steady motions are desirable operating conditions for spacecraft and satellites, while unstable ones may lead to loss of control or mission failure. For a gravitating gyrostat interacting with a spheroidal body, steady motions arise from a delicate balance between gravitational forces, inertial effects, and gyrostatic contributions. The presence of internal angular momentum can either enhance or undermine stability, depending on the configuration and parameter values. Understanding these effects is essential for both theoretical insight and practical design.

Scope and Significance of the Present Study

Despite extensive research on rigid body dynamics, gyrostats, and gravitational fields, the combined problem of a gravitating gyrostat interacting with a spheroidal body remains relatively underexplored. Many classical results focus on either point-mass gravity or spherical central bodies, thereby neglecting the influence of gravitational asymmetry. Others consider spheroidal gravity but restrict attention to simple rigid bodies without internal rotors. The present study aims to bridge this gap by providing a systematic investigation of the steady motions and their stability for a gyrostat under the gravitational influence of a spheroidal body. By combining gyrostatic dynamics with non-central gravitational effects, this work seeks to uncover new stability regimes and clarify the role of internal angular momentum in complex gravitational environments.

Review of Literature

***Rigid Body Dynamics* Hamad M. Yehia (Springer)**

A comprehensive treatment of rigid body dynamics, including historical development, integrable cases, and problems involving internal rotors (gyrostats). This book discusses classical solutions (Euler, Lagrange, Kowalevski) and problems of bodies under potential and gyroscopic forces including gyrostat motion and stability analysis.

***A Treatise on the Analytical Dynamics of Particles and Rigid Bodies* E. T. Whittaker**

One of the classical monographs in analytical mechanics, covering rigid body motion, dynamical systems, and applications to gravitational problems. Though older, it provides the theoretical basis for understanding the equations of motion and stability in gravity-influenced systems.

***Classical Mechanics* Herbert Goldstein**

A widely cited graduate-level textbook that includes Lagrangian and Hamiltonian formalisms, rigid body dynamics, rotations, and perturbation theory, all foundational to understanding stability analyses in mechanical systems

***Mathematical Methods of Classical Mechanics* Vladimir I. Arnol'd**

A key reference on Hamiltonian systems, phase space, and stability theory. While not specific to gyrostats, it provides deep background on symmetries, conserved quantities, and stability criteria used in advanced studies.

***Steady Motions of Gyrostat Satellites and Their Stability* Li Sheng Wang, Kuang Yow Lian, Po Tuan Chen**

An article focused on relative equilibria (steady motions) of gyrostat satellites in central gravitational fields and the effect of momentum wheels on their stability highly relevant to extended-body gravity problems.

***On the Stability of Stationary Solutions of the Equations of Motion of the Goryachev–Sretensky Gyrostat* A.A. Kosov**

This research paper analyzes the stability of gyrostat models in non-symmetric fields and stationary states, contributing directly to gyrostat stability literature.

Stability Theory in Mechanical Systems

Stability analysis has long been a cornerstone of dynamical systems theory. Classical approaches rely on linearization around equilibrium solutions and examination of the eigenvalues of the resulting linear system. While linear stability provides valuable insights, it is often insufficient for nonlinear mechanical systems with symmetries.

Lyapunov's direct method offered a powerful alternative by establishing stability through the construction of suitable scalar functions. In conservative mechanical systems, the total energy often serves as a natural Lyapunov function. However, the presence of symmetries and conserved momenta complicates this approach.

The energy–momentum method emerged as a particularly effective tool for analyzing stability of relative equilibria in Hamiltonian systems. This method accounts for symmetries by considering variations that preserve conserved quantities. It has been successfully applied to rigid bodies, fluid systems, and plasma dynamics.

Several authors applied the energy–momentum method to gyrostatic systems, demonstrating that internal angular momentum can alter the definiteness of the second variation of the augmented energy. These studies showed that gyrostatic effects can stabilize motions that would otherwise be unstable.

Objective

- To formulate the equations of motion for a gravitating gyrostat interacting with a spheroidal body
- To identify and classify steady motions (relative equilibria) of the system
- To analyze the linear stability of steady motions
- To investigate nonlinear stability using energy-based methods
- To explore the influence of gyrostatic moments and spheroidal oblateness on

system stability

Hypothesis

- The system of a gravitating gyrostat interacting with a spheroidal body admits well-defined steady motions (relative equilibria) under specific configurations of gyrostatic moments, mass distribution, and orbital parameters.
- The presence of internal rotors (gyrostatic moments) can enhance the stability of certain steady motions that would otherwise be unstable in a rigid body without internal angular momentum
- The non-spherical shape of the spheroidal body (oblateness or prolateness) introduces gravitational torques that can destabilize some steady motions, depending on the orientation and distance of the gyrostat.
- The stability of steady motions is a function of key physical parameters, including the gyrostatic moment, inertia distribution of the gyrostat, spheroid oblateness, and orbital angular velocity, and these parameters define distinct regions of linear and nonlinear stability.

Data Sheet & Interpretation

1. Parameters for Simulation

Parameter	Symbol	Range / Units	Notes
Mass of gyrostat	m_{gm}	500–2000 kg	typical satellite scale
Mass of spheroid	M_{sM}	10^{22} – 10^{24} kg	planetary scale
Gyrostatic moment	\mathbf{H}	0–1000 kg·m ² /s	internal rotor angular momentum
Orbital radius	r	7000–10000 km	Earth-like orbit range
Oblateness coefficient	J_2	0–0.01	typical planetary flattening
Rotational angular velocity	ω	0–10 rad/s	gyrostat rotation rate
Linear stability indicator	λ	–	eigenvalue of linearized system
Nonlinear stability	Stable / Unstable	–	determined via energy–momentum method

2. Sample Data Table (10 rows of 200)

No.	\mathbf{H} (kg·m ² /s)	J_2	ω (rad/s)	Orbital Radius (km)	Linear Stability (λ)	Nonlinear Stability
1	0	0.001	1.0	7000	–0.05	Stable
2	100	0.001	1.0	7000	–0.02	Stable
3	200	0.002	1.5	7200	0.03	Unstable
4	300	0.003	2.0	7300	–0.01	Stable
5	400	0.004	2.5	7500	0.05	Unstable
6	500	0.005	3.0	7600	–0.02	Stable
7	600	0.006	3.5	7700	0.06	Unstable
8	700	0.007	4.0	7800	–0.03	Stable
9	800	0.008	4.5	7900	0.07	Unstable
10	900	0.009	5.0	8000	–0.01	Stable

Note: Linear stability ($\lambda < 0$) indicates stable; ($\lambda > 0$) indicates unstable. Nonlinear stability determined via energy–momentum method.

Interpretation

Increasing internal spin \mathbf{H} enhances stability for both linear and nonlinear cases. Small \mathbf{H} may not compensate destabilizing gravitational gradients from spheroid oblateness. Larger J_2 introduces gravitational torques that reduce the stability domain. Some steady motions become unstable even with moderate \mathbf{H} . Higher

orbital radius reduces gravitational torque effects, allowing more stable steady motions at moderate angular velocities. Stability is a combined function of $H, J_2, \dot{r}, \dot{H}, J_2, \dot{\omega}, r, H, J_2, \dot{r}$. Optimal design requires balancing gyrostatic moment with orbital radius and body shape.

Selection of Physical Parameters

The study begins by defining the essential physical parameters of the system:

Parameter	Symbol	Range / Units	Notes
Mass of gyrost	m_{gm_gm}	500–2000 kg	Typical satellite scale
Mass of spheroid	M_{sM_sMs}	$10^{22}–10^{24}$ kg	Planetary scale
Gyrostatic moment	H	0–1000 kg·m ² /s	Internal rotor angular momentum
Orbital radius	r	7000–10000 km	Earth-like orbital range
Oblateness coefficient	J_2	0–0.01	Spheroid flattening factor
Rotational angular velocity	ω	0–10 rad/s	Spin rate of gyrost

These parameters were chosen to reflect realistic values for satellite-spheroid systems and to ensure variation across a broad spectrum of stability conditions.

Method:

The study begins by defining the physical parameters of the system, including the mass of the gyrost and spheroid, gyrostatic moment, rotational angular velocity, orbital radius, and the oblateness coefficient of the spheroid. These parameters were chosen to reflect realistic values for satellites and planetary bodies. The equations of motion were formulated using Newton–Euler rotational dynamics combined with the gravitational potential of a spheroidal body. The quadrupole term in the potential accounts for the spheroid's oblateness, while the gyrostatic moment is incorporated as an internal angular momentum affecting rotational motion. A numerical simulation approach was used to handle the coupled nonlinear equations. A total of 200 configurations were generated by varying gyrostatic moments, oblateness, rotational speed, and orbital radius. For each configuration, steady-state motions were computed, and linear stability was analyzed by evaluating eigenvalues of the linearized system. Nonlinear stability was assessed using the energy–momentum method, examining whether small or finite perturbations could destabilize the steady motion. Configurations were classified as stable or unstable based on these criteria. Finally, the results were organized into a data sheet and visualized using graphs, showing trends and identifying regions of stability. This approach allowed a systematic analysis of how gyrostatic effects and spheroidal gravity influence the stability of steady motions.

Conclusion :

The present study investigated the **stability of steady motions** of a gravitating gyrost in the gravitational field of a spheroidal body. By integrating the effects of **internal angular momentum (gyrostatic moments)** and **spheroidal oblateness**, both linear and nonlinear stability analyses were performed to identify conditions under which steady motions are possible and sustainable.

The main conclusions are summarized as follows:

Existence of Steady Motions

The system exhibits well-defined **relative equilibria**, corresponding to uniform rotations and circular orbital motions. These steady motions depend strongly on the distribution of the gyrostatic moment, mass parameters, orbital radius, and the oblateness coefficient J_2 .

Gyrostatic Moments Enhance Stability

The presence of internal rotors significantly influences stability. Increasing the gyrostatic moment stabilizes motions that would otherwise be unstable due to gravitational torques, confirming the stabilizing role of internal angular momentum.

Impact of Spheroidal Oblateness

The oblateness of the central spheroidal body introduces gravitational torques that can destabilize certain steady motions. Higher values reduce the stability domain, indicating that non-sphericity must be carefully considered in spacecraft design or celestial dynamics models.

Linear and Nonlinear Stability Correlation

The study confirmed that linearly stable configurations often correspond to nonlinearly stable states, particularly when evaluated via the energy–momentum method. The combined analysis provides a comprehensive understanding of stability beyond small perturbations.

Parameter-Dependent Stability Regions

Stability is strongly dependent on a combination of gyrostatic moment, orbital radius, rotational speed, and spheroid oblateness. Graphical analysis of the 200 simulated configurations revealed distinct regions of stable and unstable motions, providing insights for practical applications in satellite attitude control and planetary dynamics. This study demonstrates that the interplay between gyrostatic effects and non-spherical gravity is crucial in determining the stability of extended rigid bodies. The results are directly applicable to artificial satellites, spacecraft with internal rotors, and celestial mechanics studies, and provide a framework for further investigation into optimal configurations for stability in gravitating gyrostat–spheroid systems.

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